TOWARDS A MODULAR APPROACH FOR UNSTRUCTURED SHOCK-FITTING

R. Pepe¹, A. Bonfiglioli¹, R. Paciorri², A. Lani³, J. G. Mena³ C. F. Olliver-Gooch⁴

¹ Università della Basilicata , 85100, Potenza, Italy aldo.bonfiglioli@unibas.it
² Università di Roma, 00184, Roma, Italy, renato.paciorri@uniroma1,it
³ von Karman Institute, 1640 Sint-Genesius-Rode, Belgium, lani@vki.ac.be
⁴ University of British Columbia, V6T 1Z4, Vancouver, Canada, cfog@mech.ubc.ca

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The *unstructured*-grid version of the shock-fitting technique made its first appearance in 2009 [1] and has been developed since then: it has been successfully applied to 2D steady flows featuring interacting shocks [2] and 3D steady flows with multiple, though not interacting shocks [3].

Unstructured shock-fitting retains features of both the "boundary" (algorithmically simple) and "floating" (topologically more versatile) variants of the algorithm that had been used in the structured grid setting: the fitted shock fronts are interior *boundaries* that are free to *float* over a background mesh which is locally re-meshed to adapt to the actual position of the shock front; a shock-capturing solver is used to discretize the governing PDEs in smooth regions of the flowfield.

The use of unstructured grids allows to relieve most of the algorithmic difficulties encountered in the structured grid setting whilst keeping the key feature of the "traditional" shockfitting discretization, i.e. accuracy on coarse meshes.

Modularity is one of the key features of this newly developed unstructured shock-fitting algorithm. Indeed, since the fitted shock fronts behave as interior boundaries, the CFD solver can be used as a "black box"by the shock fitting algorithm and the same consideration applies to the algorithm that is used for local re-meshing around the shock-front. For instance, the technique had originally been developed using the in-house eulfs CFD solver [4] and different public domain codes for volumetric and surface remeshing, such as: triangle [5] in 2D and tetgen [6] and Yams [7] in 3D. Very recently, the CFD solver has been replaced with the COOLFluiD[8, 9] code developed at VKI. Thanks to the high modularity of the shock-fitting technique, this task has been accomplished at a very limited coding effort. Figure 1, which refers to the supersonic flow past a circular cylinder, shows pressure iso-contours computed by CoolFluid in shock-capturing and shock-fitting mode on the same coarse mesh. The comparison clearly highlights the improvements in solution quality that shock-fitting enables.



Figure 1: Supersonic flow past a 2D circular cylinder

In the near future, we plan to use the local remeshing technique specifically designed by Zaide and Olliver-Gooch [10] to remesh the shock region. This will allow to improve the performance of the shock-fitting technique and to broaden its range of application, especially in the field of 3D flows.

In the full paper the integration of the shock-fitting algorithm with the COOLFluid solver and the new mesh generation subroutines will be discussed and more examples of flow computations will be shown.

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