

ADAPTATION OF 2D UNSTRUCTURED MESH BASED ON SOLUTION GRADIENTS

R. Vijay Ram^{1*}, Santanu Ghosh², Shashank Subramanian³ and Deepak Kandasamy

¹ IIT Madras, Dept. of Aerospace Engineering IITM, vijayram.ae@gmail.com
² IIT Madras, Dept. of Aerospace Engineering IITM, sghosh1@iitm.ac.in
³ IIT Madras, Dept. of Aerospace Engineering IITM, shashank1693@gmail.com
⁴ IIT Madras, Dept. of Aerospace Engineering IITM, deepak.vk.1994@gmail.com

Key words: *Mesh Adaptation, Unstructured Mesh, Gradient Based Methods*

The work presented in this paper attempts to improve the grid (and consequently solution) for simulations on a 2D unstructured Navier-Stokes solver for compressible flows using mesh adaptation based on gradients of flow parameters (pressure and pseudo-entropy) and grid geometry (cell areas, nodal distances etc.). A formula was developed to move nodes based on the adaptive techniques discussed in Eiseman [1] and Jahangirian and Shoraka [2]. The procedure requires solution (primitive variables) reconstruction at nodes, which are interpolated using Lagrangian polynomials or inverse distance based methods, and cell-averaged gradients, which are computed using Green-Gauss method. The adaption is terminated if the global maximum displacement of any node is less than ϵ , where ϵ is a small user defined number. A 1D version of this formula is tested on the Heaviside function $H(x)$ discretized on the domain $x \in [-1, 1]$ using 11 points. The results obtained are shown in Figure 1. All points clustered about $x = 0$ as expected.

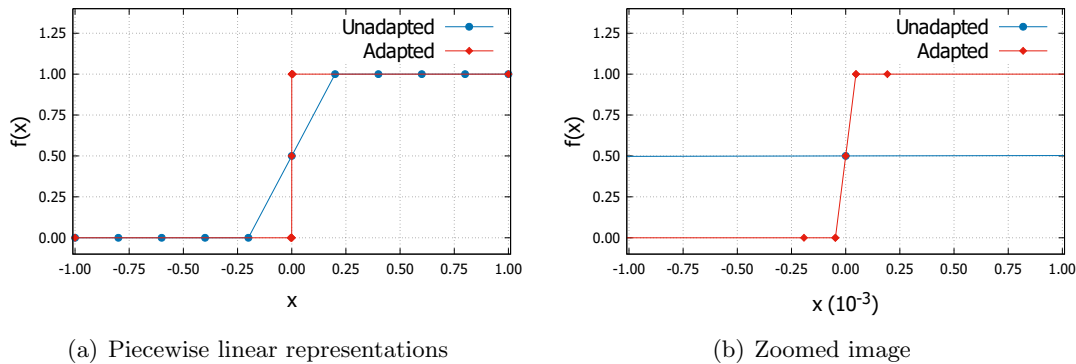


Figure 1: 1D adaptation of the Heaviside function

REFERENCES

- [1] Eiseman, R. Peter, Adaptive Grid Generation, *Computer Methods in Applied Mechanics and Engineering*, 321-376, 1987.
- [2] Jahangirian, A. and Shoraka, Y., Adaptive unstructured grid generation for engineering computation of aerodynamic flows, *Mathematics and Computers in Simulation*, **78**, pp. 627-644, 2008.